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New Material Transition From an OEM Perspective

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Agenda



In the composites industry, there is significant, widespread confusion and frustration relating to material transition onto products. This presentation goes into several areas pertaining to an increased understanding from a multiple discipline perspective.

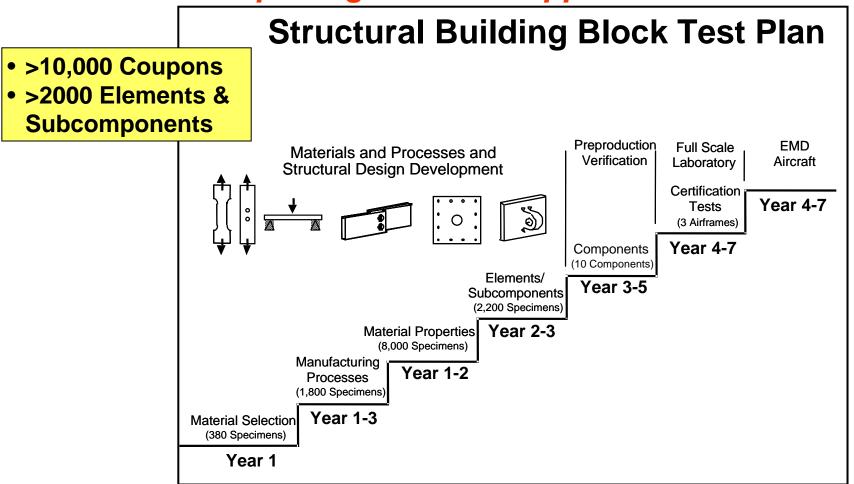
- Background
- Maturity
- Disciplines
- Conformance
- Summary



Background



Transition means putting it onto an application...



...Application Acceptance is Through Certification With Stepwise Risk Reduction



Background



Designer Data/Information Needs for Application Certification

Structural

- Strength and Stiffness
- Weight
- Service Environment
 - Temperature
 - Moisture
 - Acoustic
 - Chemical
- Fatigue and Corrosion Resistant
- Loads & Allowables

Manufacturing

- Recurring Cost, Cycle Time, and Quality
- Use Common Mfg.
 Equipment and Tooling
- Process Control
- Inspectable
- Machinable
- Automatable
- Impact on Assembly

Supportability

- O&S Cost and Readiness
- Damage Tolerance
- Inspectable on Aircraft
- Repairable
- Maintainable
 - Accessibility
 - Depaint/Repaint
 - Reseal
 - Corrosion Removal
- Logistical Impact

Material & Processes

- Development Cost
- Feasible Processing Temperature and Pressure
- Process Limitations
- Safety/Environmental Impact
- Useful Product Forms
- Raw Material Cost
- Availability
- Consistency

Miscellaneous

- Observables
- EMI/Lightning Strike
- Supplier Base
- Applications History
- Certification Agency & Status
 - USN
 - USAF
 - ARMY
 - FAA

Risk in Each Area is Dependent Upon Application's Criticality and Material's Likelihood of Failure





Technology Readiness Level (TRL) For Maturity

Designer Perspective

Technology Readiness Level

System

- 10. Disposal
 - 9. Production
 - 8. Flight Test
 - 7. Ground Test
 - 6. Component Test
 - 5. Design Maturation (Subcomponents)
- 4. Preliminary Design
 (Stable Mat'l & Process
 + Elements)
- 3. Proof of Concept Prototype
- 2. Concept Definition
- 1. Concept Exploration

PROS

- Looks at maturity from a designer/system viewpoint
- Broken down into specific activity areas
- Is geared towards application products and systems for readiness

CONS

- Does not take into account different discipline perspectives
- Does not address detailed areas/items at each readiness level

Based on NASA, Air Force and ONR Technology Maturity Level Approaches





Readiness Levels From a Technologist Viewpoint

Technologist Perspectives

Readiness Level

- 9. Industry Std
- 8. Production
- 7. Qualified Mat'l/Process/Mfg
- 6. Pre-Production
- 5. Pilot Production
- 4. Lab/Prototype Production
- 3. Beaker/Bench Product
- 2. Theoretical/Beaker Product
- 1. Concept Exploration

PROS

- Looks at maturity from a technologist viewpoint
- Broken down into specific activity areas
- Is geared towards materials, processing and manufacturing for readiness

CONS

- Is not tied/connected to TRL's from the system or application viewpoint
- Does not take into account different discipline perspectives
- Does not address detailed areas/items at each readiness level





Certification

- Application Requirements/Needs Demonstrated
- Stepwise Risk Reduction (Building Block Approach)
- Performance Characteristics for Primary/Secondary/ Air Loaded Structures

Qualification

- Materials and Processing are Stable
- Material and Processing Specifications

Transition

 Customer Acceptance for an Application or Applications





Connections/Correlations for Readiness Levels

Technologist Perspectives Designer Perspective **Technology** Readiness Level Readiness Level 10. Disposal 9. Industry Std 9. Production 8. Production System 8. Flight Test 7. Ground Test 6. Component Test **Technologist** Activity 5. Design Maturation Description (Subcomponents) 4. Preliminary Design ← 7. Qualified Mat'l/Process **Final Capabilities** (Stable Mat'l & Process + Elements) 3. Proof of Concept **Expanded Capabilities Prototype** 2. Concept Definition

5. Pilot Production **Preliminary Capabilities** 1. Concept Exploration

4. Lab/Prototype Production **Preliminary** 3. Beaker/Bench Product Investigations, Activity Steps Moving to 2. Theoretical/Beaker Product Research. Certification Development 1. Concept Exploration

Activity Steps Moving to Qualification



Multiple Disciplines



TRL	0	1	2	3	4	5	6	7	8	9	10
Certification	Qualification Plan Assessment	Certification Elements Documented	Certification Plan Documented	Certification Plan Approved	Elements	Subcomponent Testing	Full Scale Component Testing	Full Scale Airframe Tests	Flight Test	Production Approval	Disposal Plan Approval
Application/ Design	Technology Readiness Review	System Requirements Review	Project Planning Review	Preliminary Design Review	Critical Design Review	Full Scale Test Readiness Review	Ground Test Certification Review	Flight Test Certification Review	Production Readiness Review	Production Support	Recycle or Dispose
Assembly/ Quality	Preliminary Assembly Concept Assessed	Assembly Concept	Assembly Plan Definition	Key Assembly Detail Definitions	Key Assembly Details Tested	Subcomponents Assembled	Components Assembled	Airframe Assembled	Flight Vehicles Assembled	Production	Disassembly for Disposal
Survivability	General Requirements Assessed	Requirements Definition	Concept Definition	Proof of Concept	Preliminary Design Data and Guidelines	Design Allowables and Guidelines Defined	Critical Details Testing	Ground Test	Flight Test	Production Support	Operations Support & Disposal
Fabrication/ Quality	Fabrication Capability Demonstrated	Unfeatured-Panel Fabrication	Feature Based Generic Small/Subscale Parts Fabricated	Property-Fab Relationships Tested/ Target Application Pilot Production of Generic Full Size Parts	Process Specs/ Effects of Fab Variations Tested/ Elements Fab'd/ Production Representative Parts Fab'd	Subcomponents Fab'd	Full Scale Components Fabricated	EMD Fabrication	Low Rate Initial Production (LRIP)	Production	Recycle or Disposal
Supportability	Repair Requirements Assessed	Repair Items/Areas Identified	Repair Materials & Processes Identified	Repair Materials & Processes Documented	Fab Repairs Identified	Fab Repair Trials/ Subcomponent Repairs	Component Repairs	Production Repairs Identified	Flight Qualified Repairs Documented	Repair-Replace Decisions	Support for Recycle or Disposal Decisions
Structures & Durability	Preliminary Properties- Characteristics Assessed	Preliminary Properties- Characteristics	Initial Properties	Design To Properties Developed	Preliminary Design Allowables	Final Design Allowables	Allowables for Critical Design Features	Production and Test Support	Certified Allowables	Flight Tracking/ Production Support/ Fleet Support	Retirement for Cause
Materials	Lab-Prototype Materials	Lab-Prototype Materials Reproducibile	Pilot Production Materials	Pre-Production Materials	Production Sacleability Validated	EMD Material Supplied	EMD Material Supplied	EMD Material Supplied	LRIP Material Supplied	Production Material Supplied	Support for Recycle or Disposal Decisions
Cost/Schedule/ Benefits	Cost Benefit Elements ID'd & Assessed	Cost Benefit Elements ID'd & Projected	ROM Cost Benefit Analysis	Cost Benefit Analysis Reflect Size Lessons Learned	Cost Benefit Analysis Reflect Element and Production Representative Part Lessons Learned	Cost Benefit Analysis Reflect Subcomponent Fab & Assembly Lessons Learned	Cost Benefit Analysis Reflect Component Fab & Assembly Lessons Learned	Cost Benefit Analysis Reflect EMD Lessons Learned	Cost Benefit Analysis Reflect LRIP Lessons Learned	Cost Benefit Analysis Reflect Production Lessons Learned	Cost Benefit Analysis Reflect Disposal Lessons Learned
Intellectual Rights	Concept Protection Plan Developed	Protection Plan Documentation	Patent Disclosure Filed	Proprietary Rights Agreements	Data Sharing Rights	Vendor Agreements	Material and Fabrication Contracts	Production Rate Contracts	Vendor Requal Agreements	Post-Production Agreements	Liability Termination Agreements

...Multiple Disciplines Have Different Perspectives for Technical Maturity Level Exit Criteria



Multiple Disciplines



Structures and Durability Breakout

Exit Criteria Vary According to Application

Дррііч	Sation
In-plane Ultimate Strength Unnotched Compression	Bond/Interlaminar Joint Strength - Final Failure
In-plane Ultimate Strength Unnotched Tension	Bolted Joint -
Ultimate Strength Combined Loads	Bearing/Bypass
In-plane Ultimate Strength Open Hole Compression	Maximum Deflection
In-plane Ultimate Strength Open Hole Tension	
Ultimate Strength Open Hole, Combined Loads	Residual Strength BVID, Compression
Stability - Global/Panel	Residual Strength Penetrations, Tension
	Residual Strength Penetrations, Compression
Stability - Skin Buckling	Residual Strength Penetrations, Combined Loads
Stability - Stringer Crippling	Local Stability - Face wrinkling (Sandwich Only)
Stability - Stringer Column Buckling	Local Stability - Intracell Buckling (Sandwich Only)
Bond/Interlaminar Joint Strength - Damage Initiation	Local Stability - Shear Crimping (Sandwich Only)

Exit Criteria Tends to Vary According to OEM and Customer

Durability/Life Microcracking

Durability/Life Delamination Growth

Durability/Life - Stiffness
Degradation

Durability/Life - Bearing
Strength Degradation

Material Mechanical Properties - Primary (Tension, Compression, Shear, Bearing By-pass) **Material Mechanical** Properties - Secondary (CTE, Poisson's, Fracture Toughness,) **Material Mechanical** Properties - Other () Material Durability/Life Properties -Environmental Impact on Properties Material Durability/Life Properties - Impact Resistance and Fatigue Material Durability/Life Properties - Solvent Resistance



Multiple Disciplines



Materials, Processing and Producibility Breakout With Critical quality functions/characteristics Identificate material and/or Critical quality functions/characteristics Ident

MATERIAL	Critical functions/ characteristics of material/ ingredients demonstrated. New material within state-of-the-art. Indirect material requirements identified. Facility requirements identified.
PROCESSES	Critical functions/ characteristics of processing demonstrated. New process operates within state-of-the-art. Facility requirements identified. Indirect materials or process steps identified.
EQUIPMENT	Critical functions/ characteristics of individual equipment pieces demonstrated. Indirect materials and facility requirements identified. Equipment accuracy requirements defined.
TOOLING	Critical functions/ characteristics of individual tooling pieces demonstrated. Indirect materials and facility requirements identified. Tooling accuracy requirements defined.
VARIABILITY	Variabilities roughly characterized.

QUALITY - IN- PROCESS	Critical quality functions/characteristics demonstrated. Indirect material and/or process steps identified. Facility requirements identified. Defects identified
QUALITY - FINAL PRODUCT	Critical quality functions/characteristics demonstrated. Indirect material and/or process steps identified. Facility requirements identified. Defects identified.
APPLICATION MATURITY	Critical functions/characteristics demonstrated; physical phenomena understood.
COST/BENEFIT ANALYSIS	Key costs/benfits have had a preliminary assessment for quantification.
SUPPORTABILITY	Critical repair functions and characteristics demonstrated.
REGULATORY	Regulatory issues understood.
Intellectual Property	Reduction to practice in progress. Strategy to issue patents or preserve technology as trade secret accepted.

Combines Both Objective and Subjective Areas

- Includes Property and Characteristic Measurements
- Includes Production Readiness Assessments
- Each Individual Material and Processing/ Producibility Step Needs to be Addressed



Conformance



......Varies According to Application, Maturity, Discipline and Customer To Meet Requirements

			(Quasi)	Longitudinal	Fille	amin	ate						•		
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				Longitudinal		e Tensile St	rength					, .			
			Layup 2		Filled Hol			Viscosity				/iscosity			
	_		(Hard)	_			rain to Failur	Reaction Rate				Degree of Cure			
	· ·	ension		Transverse	Open Hol	e Tensile St	trength	Heat of Reaction				HPLC		\neg	
		Longitudinal	Strength	, Modulus, Str	ain to Failu	ire Poisson	s 1 & 2	Volatile Cont	mpera	ature	F	TIR	Prepreg		
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Tension							volatile vapol			F	Resin Areal Weig	nht .			
		Through	Strength	∣Lam	ina			Resin Cost				iber Areal Weig	,		
		Thickness	Modulus					Density Resin Cure Shrinkage	^			Mass Fraction Fil			
		Longitudinal	Strength.	, Modulus, Str	ain to Failu	ure, Poisson:	s1&2	CTE				lass Flaction Fil			
Compression		Transverse	Strength,	, Modulus, Str	ain to Failu	ure, Poisson:	s1&2	Thermal Conductivity		Tensile Strength		_	city		
Joinpression		Through	Strength	n				Specific Heat		Tensile Modulu		u <mark>dinal)</mark>			
		Thickness						Kinetics Model		Tensile Strain	- Iber		oer		
			Strength					Viscosity Model		AC - Lat (NALIL)			kress		
No		Longitudinal	Modulus					Intellectual Property I	Issues	Yield (MUL)			tht		
Shear	. 450		Modulus 2 Strength, Modulus, Strain to Failure					HPLC					ility, x		
	±45°	lu e			ain to Fall	ure		FTIR		Heat Capacity (<u> </u>	ility, y			
	U°	ILS	Strength Bulk Mod					Health and Safety Inf			<i>-</i> ρ)				
					ratio 2			Morphology		Thermal Conduc		Requirements	.,		
							Ingredient Suppliers		Thermal Conduc		Spool Informa				
				Thickness Th		ductivity		Cured Resin					ials ID/Compatibility		
			Specific Heat				Tensile Stress to Fail		CTE - Axial						
		Longitudinal		11041				Young's Modulus, Te				Tack, Original			
		Transverse						Tensile Strain to Faile		CTE - Radial Han	Hand Cutti	Tack, Out Tim			
Other		Through	Coeff. Of	f Thermal Exp	pansion			Glass Transition Tem	perature		Hand Cutti	rack, Freezer			
		Thickness						Volatile Content		Filament Diame	r	Variability, Dir			
		Longitudinal						Density Madulus as a Function		Filament Count	u i	Variability, An	gle		
		Transverse	Coeff Of	f Moisture Exp	nancion			Modulus as a Function			-		Draft Items/Areas		
		Through	Coeii. Oi	i Moisture Exp	Jansion			Thermal Conductivity		Transverse Bulk	. VI	Proce	ssing/		
		Thickness						Solvent Resistance		Youngs Modulus	S				
		Longitudinal	Thermal	Conductivity				Specific Heat		Shear Modulus,	6	Produ	cibility		
		Transverse						Bulk Modulus		Shear Modulus,	6	1	/C p ility		
			(Soft)	Transverse		1, Modulus 1		Shear Modulus		Poissons Ratio,	2	Tack Original	(lay down and removal)		
			` /			2, Modulus 2		Poisson's Ratio		Poissons Ratio,	1 3		e (lay down and removal)		
			Layup 1			Strain to Fai		Coefficient of Moistur		Compressive St		Took Fronzer			
		Pooring Dy	(Quasi)	Transverse		Strain to Fai		Compression Strengt		Cost	Hand Layu	Variability, Dir			
		Bearing By- Pass, Tension	Layup 2	Longitudinal		Strain to Fai		Compression Modulu	IS .	Tg)		Variability, An			
		ass, rension	(Hard) Layup 3	Transverse		Strain to Fai Strain to Fai		Mass Transfer Prope	11165		-		Draft Items/Areas		
			(Soft)	Transverse		Strain to Fai		Viscoelastic Propertie	50	wet Tg)					
			Layup 1			Strain to Fai		Toughness Properties		Health and Safe	4	Specification, Specification,			
			(Quasi)	Transverse		Strain to Fai		Tg, Wet		Fiber Surface		Specification,	FIIIdl		
		Bearing By-	Layup 2			Strain to Fai		CME	(Sizing Type					
		Pass,	(Hard)	Transverse		Strain to Fai		Solvent (Moisture) Di		Fiber Surface Ro	oughness				
	C	Compression	Layup 3	Longitudinal		Strain to Fai		Solvent Resistance	_	Surface Chemis					
			(Soft)	Transverse		Strain to Fai				Carrace Criciilis	uy		BOEI		

Summary



- Multiple Discipline Perspective Integrates Requirements and Conformance to Requirements
- Multiple Discipline Conformance Activities Covers The Spectrum of All Areas and Items for Material Insertion
- Certification and Qualification is Very Complex
- Primary Emphasis is Risk Reduction Relative to Applications



Ending



"The best information on a new material are the first things heard about it. It only goes downhill from there."

Quote Attributed to Flake Campbell Jr.



